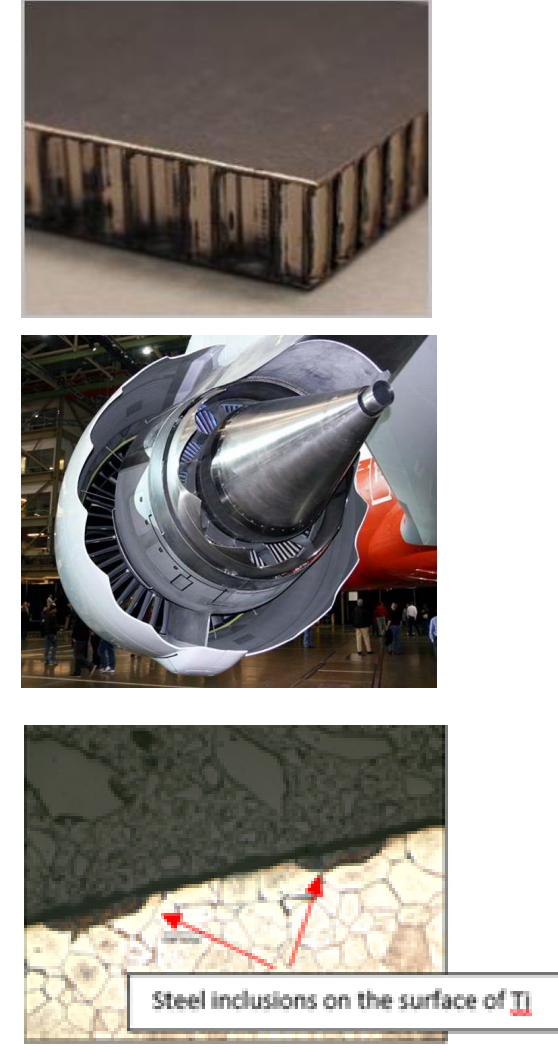


Introduction



Background:

Certification of new parts requires prediction of fatigue life of the component. GKN Aerospace currently uses an overly conservative method to assess certain manufacturing defects and seeks a more reliable tool.

Motivation:

GKN Aerospace produces engine exhausts (Titanium Honeycomb formed at elevated temperatures). As part of this process, inclusions of dissimilar materials can be experienced. GKN seeks more efficient method to predict reliability of parts with embedment of dissimilar metals

Problem Statement:

Obtain understanding of effect of embedded dissimilar materials observed in GKN Aerospace manufacturing process, and *deliver an analytical method for its prediction on fatigue life.*

Customer Specification:

- Based on existing regulatory guidelines, develop analytic methodology to predict life capability impact of embedded defects on component performance
- Primary focus on tensile loading
- The methodology should be easy to use and reasonably accurate for design purposes.
- Material specified as Ti-CP-Gde4

Analysis

Considerations:

After reviewing 14.CFR.25/14.CFR.33, considering material features and damage tolerance guidelines, it is concluded regulatory agencies place more restrictive requirements on fatigue loading and fatigue life, hence this should be the primary focus

Use a analytic method to produce reliable results at low computational complexity for design purposes

After reviewing NASA papers and consulting faculty advisor Prof. M. Salviato, it is determined this method has merit, could be adopted to solving the problem, and could produce result with greatest computational efficiency of all methods considered

Decision:

- Primary focus on fatigue life under cyclic loading
- Use finite element analysis (FEA)
- Validate P.Lazzrin proposed method on given problem

Budget

On budget, with surplus

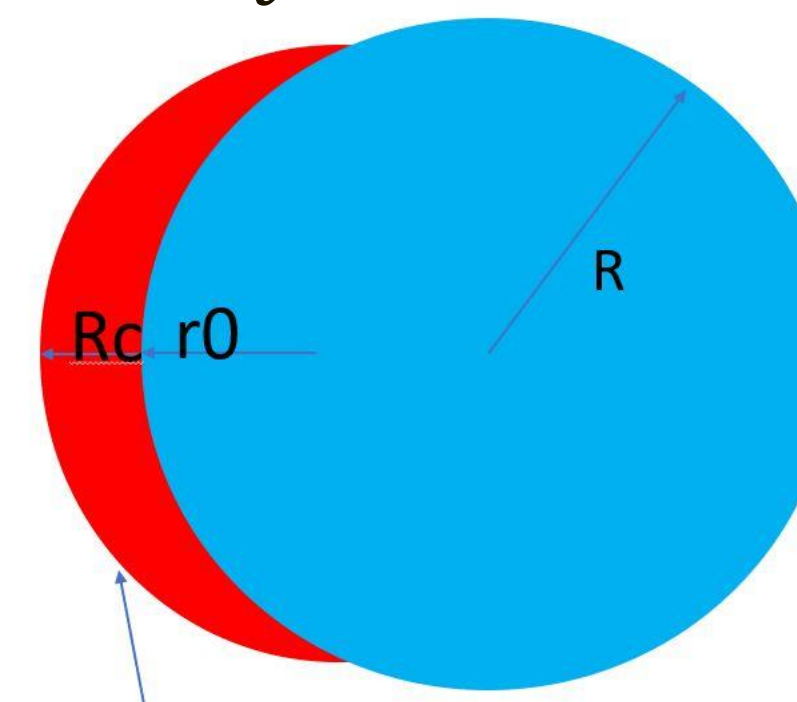
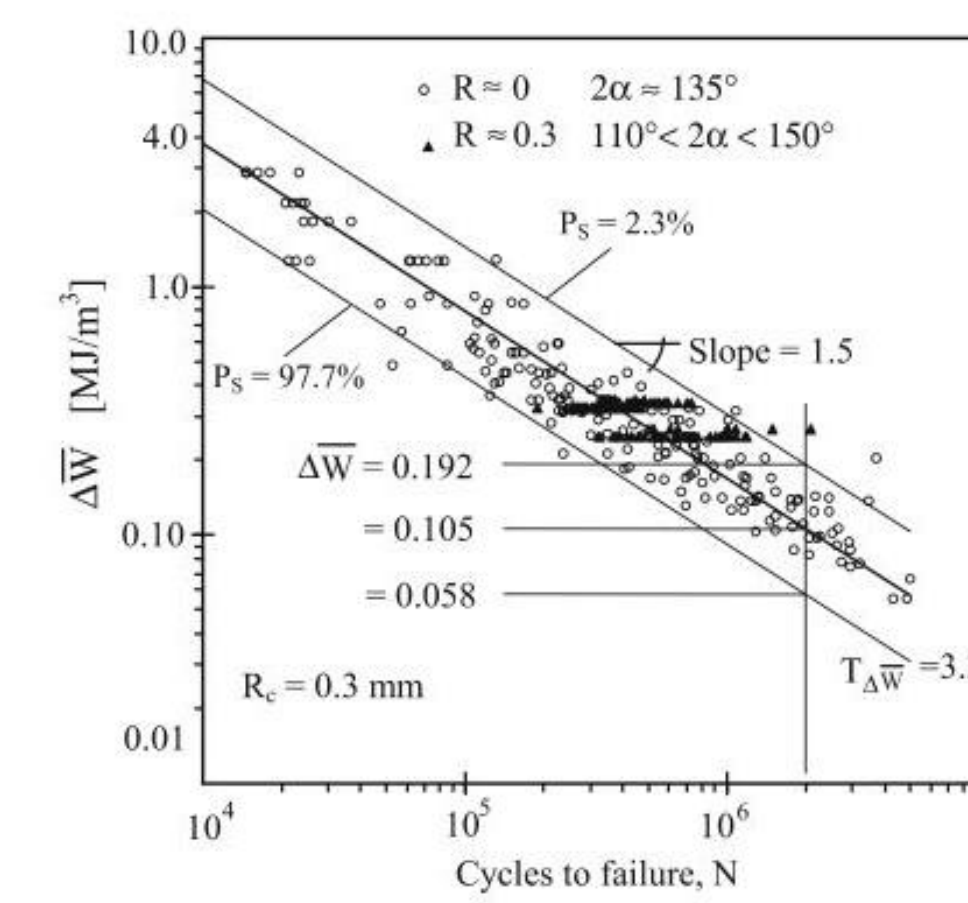
Item	Planned	Actual
Additional Testing Material / Buffer	\$760-\$2200	\$2,600
Commercial Testing Service / Buffer	\$1,440	\$0
Specimen Manufacture	\$1,800	\$1,100
Total Budget	Actual Spent	Remaining
\$4,000	\$3,700	\$300

Detail Design & Workflow

Lazzrin Method:^[1,2]

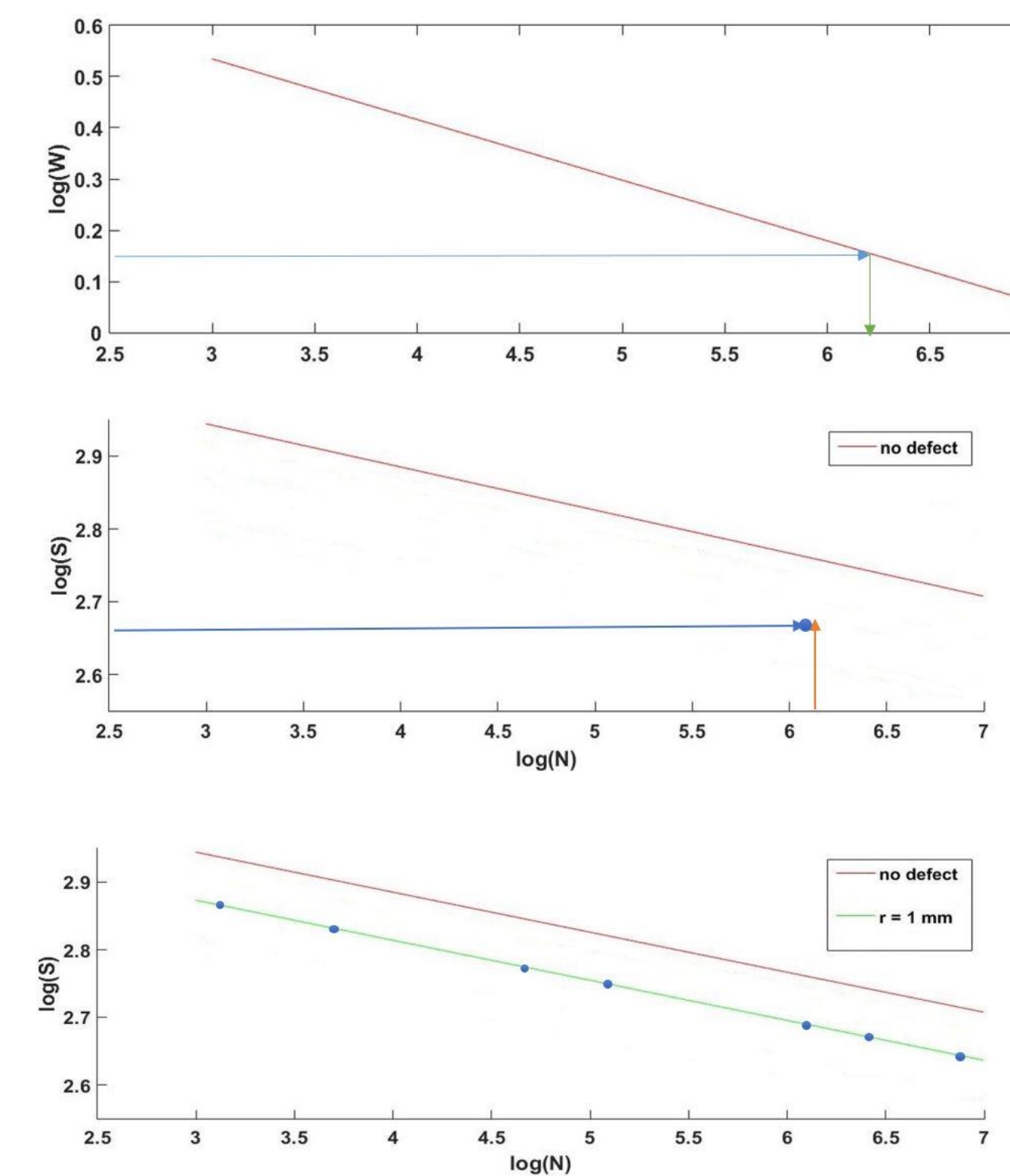
- Control volume determined by R_C
- Calculated entirely from material properties
- Average Strain Energy Density linearly related to cycles in log-log domain
- Works with geometries where stress varies throughout material

$$R_C = \frac{(1 + \vartheta)(5 - 8\vartheta)}{4\pi} \left(\frac{K_{IC}}{\sigma_t} \right)^2$$



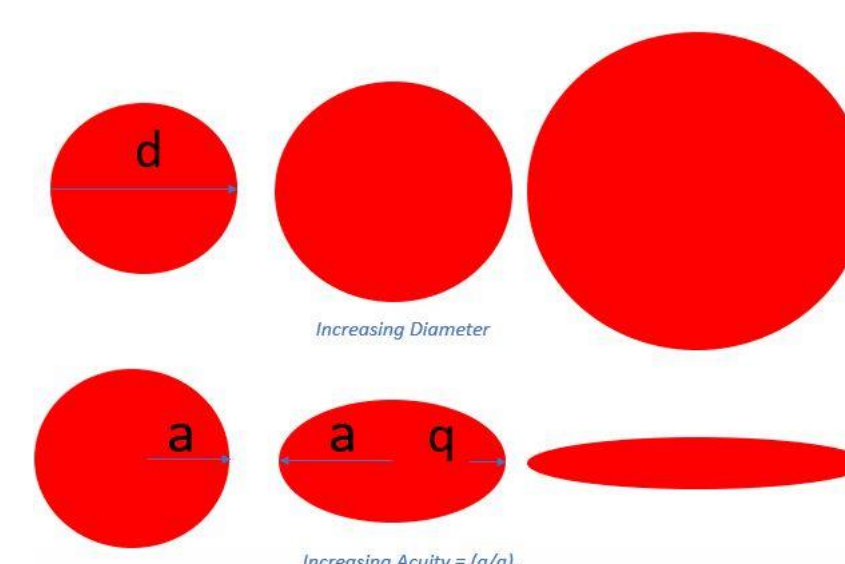
Characteristic volume

- For a given loading and geometry, FEA can be used to derive average strain energy density, W , throughout plate
- W^*V summed for elements in control volume, divided by volume of control volume
- W then used to get N from $W-N$ curve
- Can then be used to construct an $S-N$ curve for the geometry

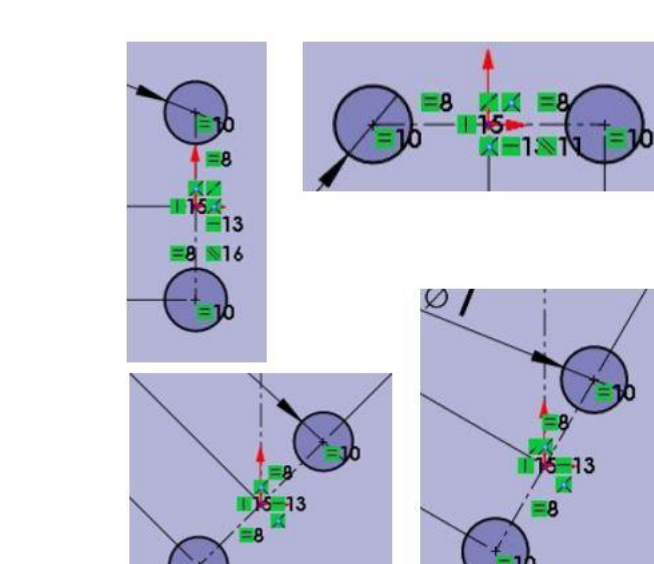


Validation:

- Perform tests to determine GKN material prosperities
 - ASTM E8, E466, E1820
- Utilize material property to determine critical defect size
- Engineer through-all defects to specimens
 - Size, Acuity, Relation
- Compare FE model prediction and actual testing result and resolve discrepancies
- Validate calibrated model with real defect



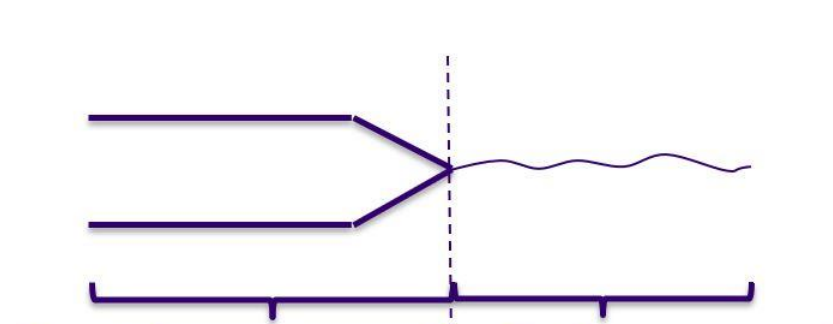
Acuity Defect Arrangement Design



Relationship Defect Arrangement Design



DIC Fracture Testing Specimen



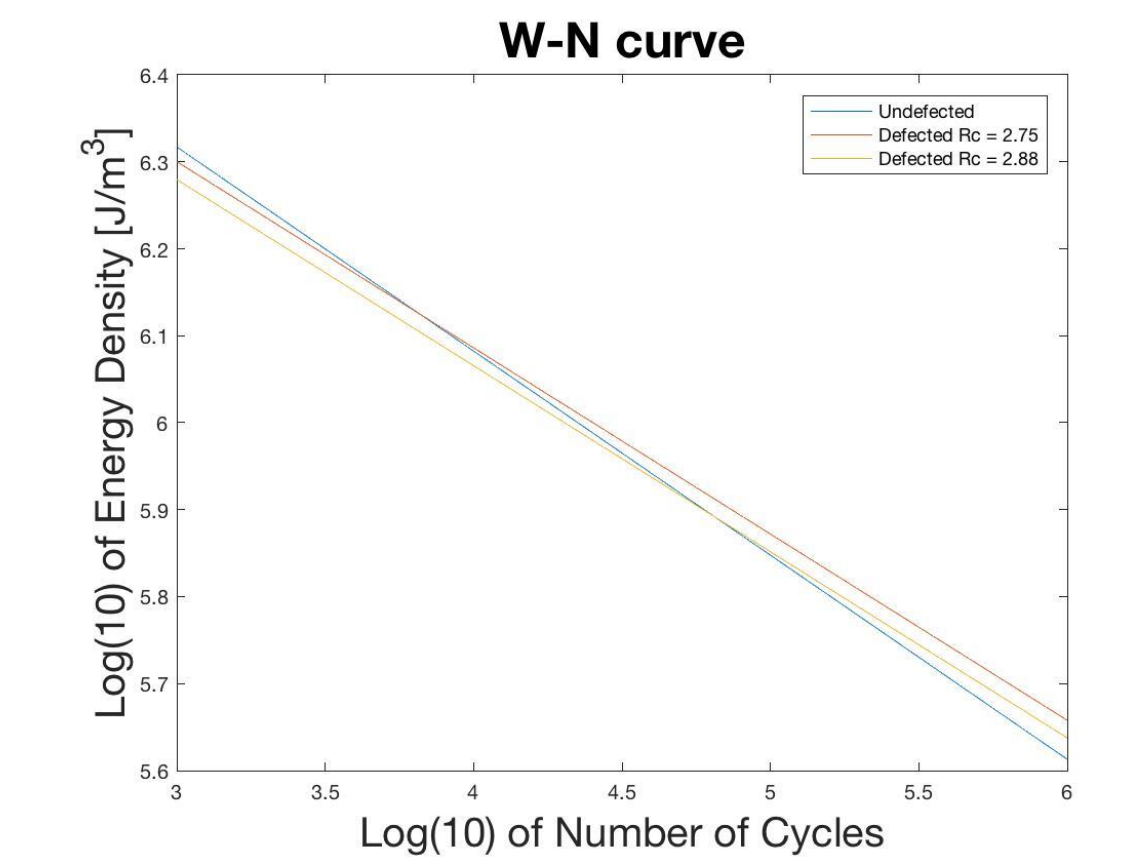
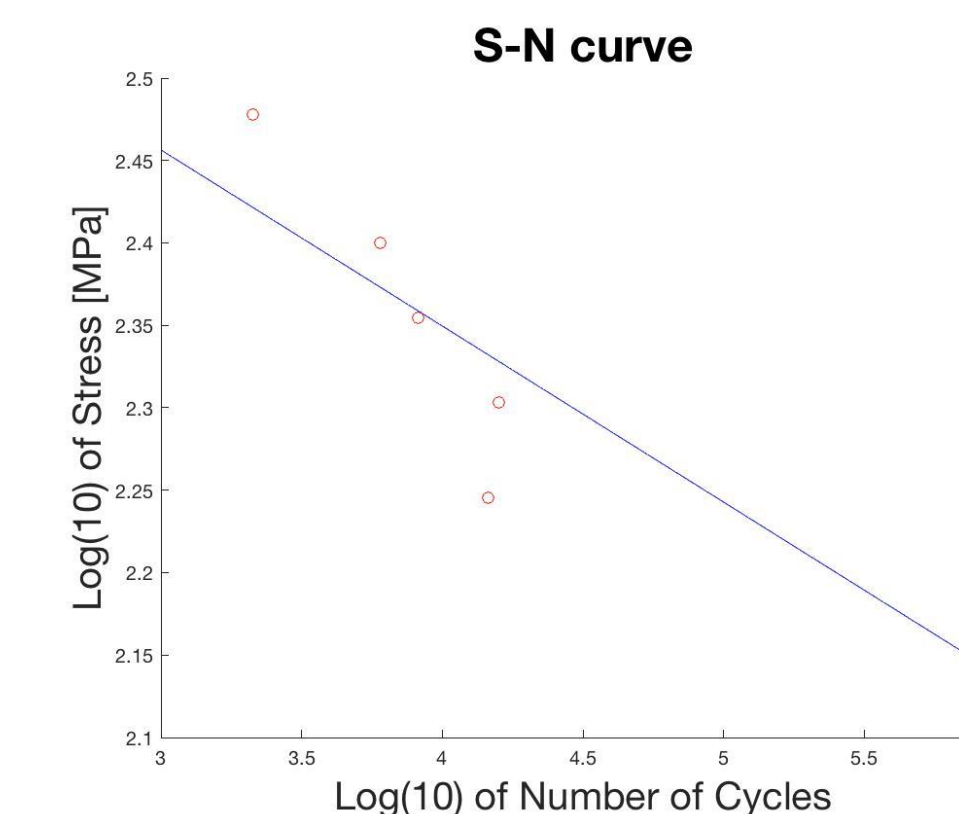
Fracture Test Notch Design

Schedule

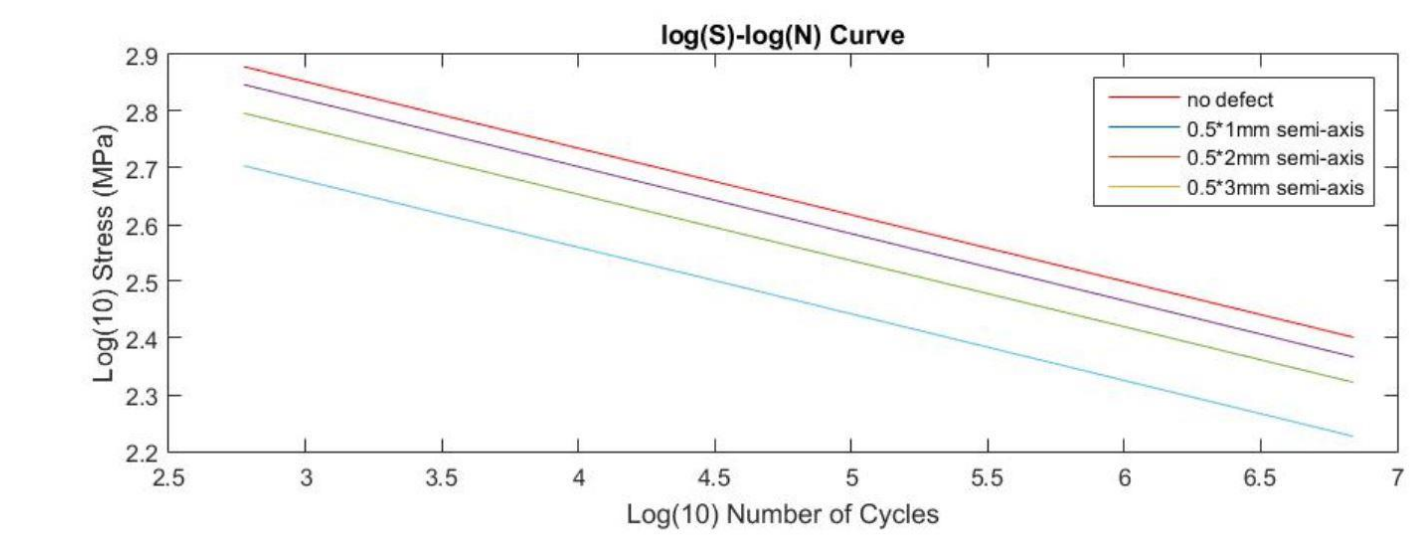
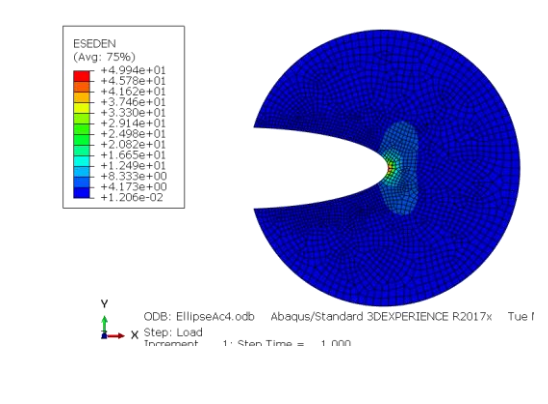
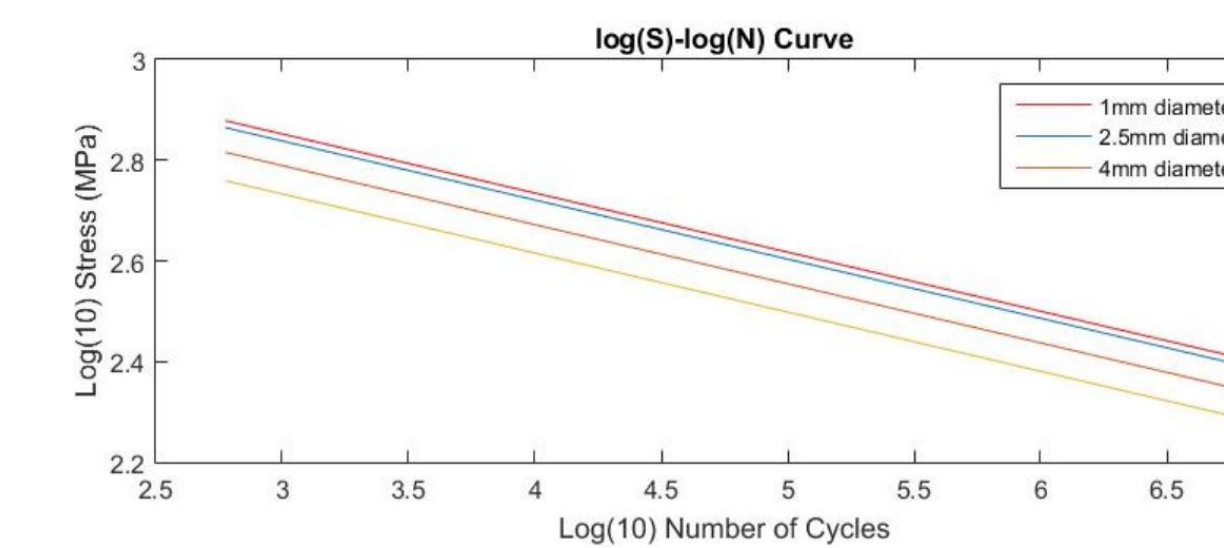
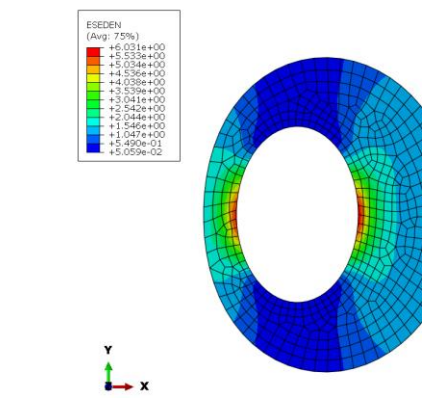
Month	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	
Task / Week#																					
Initial Design																					
Construct FEA Model																					
Obtain Testing Material																					
Detailed Design																					
Perform Calibration Test																					
Calibrate Model w/ Result																					
Engineer Defected Specimens																					
Perform 2nd Test																					
Consolidate Result																					
Project Buffer																					
Testing Buffer																					
Analysis Buffer																					
Project Close-up																					

Key Results

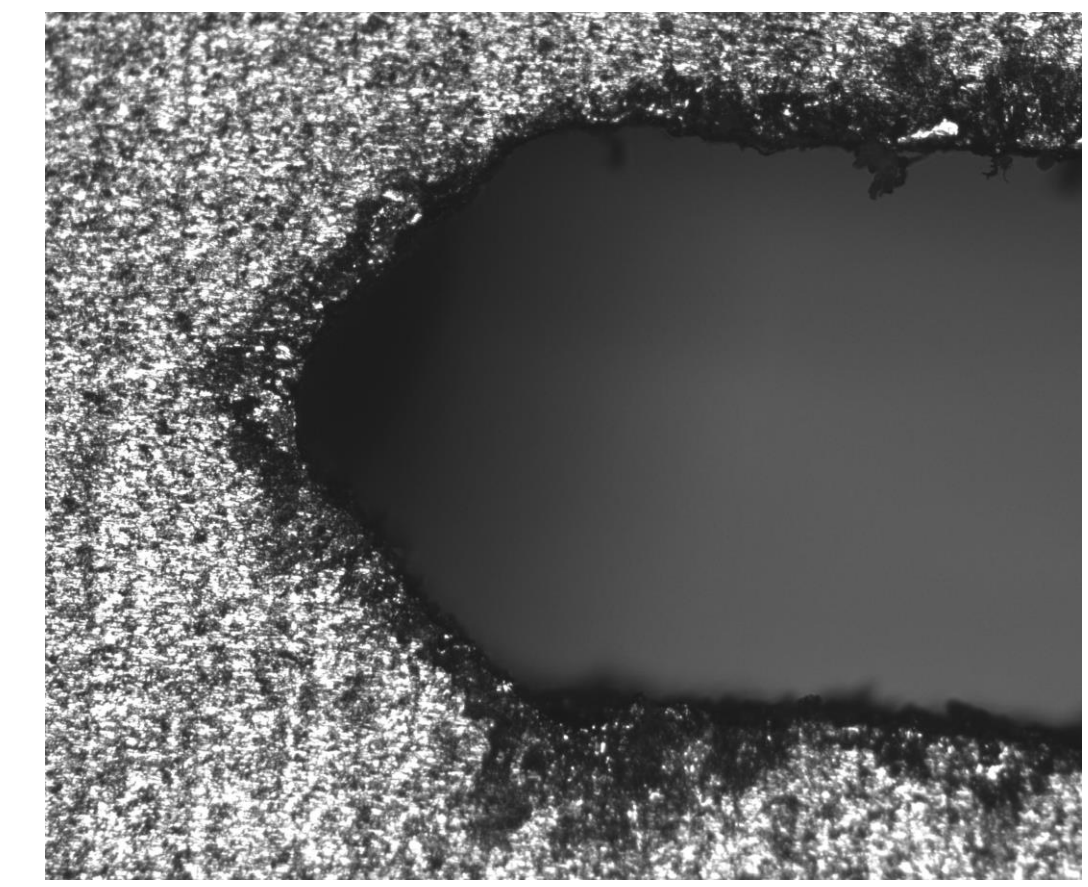
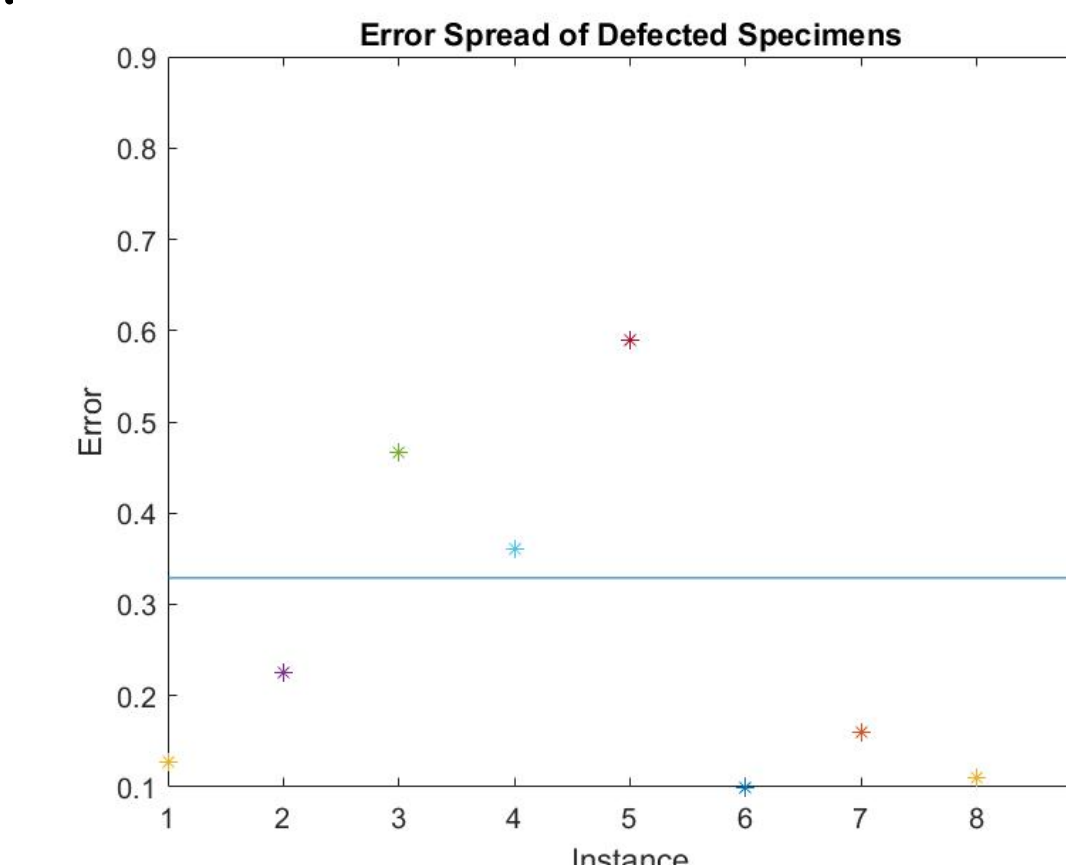
- Lazzrin Method uses quasi-static tests to obtain R_C and is only thus far validated on brittle metals. When applying the method to ductile material such as Ti-CP-Gde4, the model prediction offers a median error of 42% comparing to experimental method.



- Further research indicates using quasi-static method to predict fatigue behavior tend to yield erroneous results, and R_C was recalibrated with another method recommended by Prof. M. Salviato.



- The new calibrated R_C yield around 32% error, considering the scatter factor, spread pattern of data, defects possibly introduced to the specimen during defect manufacture, and nature of fatigue testing, the result indicate Lazzrin Method could be reasonably adopted for this research.
- The project developed and delivered a Finite Element model that shows promisingly accurate and less conservative result then current practice produced at higher computational efficiency.



Microscopic Inspection of Defected Specimen

Future Work

- Perform more test with the same material and same procedures to establish statistical significance
- Extend verification of the Lazzrin method to a wider range of metal kinds and specimen parameters
 - Ductility, depth, metal bound (Unable to investigate since no proprietary specimens provided by GKN)
- Compare extended verification results with industry records, make adjustments accordingly and promote the final result for wider adoption

Acknowledgement

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Prof. M. Salviato, for guiding the team with exceptional theoretical knowledge technical knowhow
Prof. Morgansen, Prof. Waas, ME shop, Physics shop, and PhD candidate Yao Qiao for facilitating the development of the project



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